Supersonic Flow Over a Wedge

A uniform supersonic stream encounters a wedge with a half-angle of 15 degrees as shown in the figure below.

The stream is at the following conditions:

\[
\text{Mach Number } M_1 = 3
\]
\[
\text{Static Pressure } p_1 = 1 \text{ atm}
\]
\[
\text{Static Temperature } T_1 = 300 \text{ k}
\]

Using FLUENT, calculate the Mach Number, static and total pressure behind the oblique shock that will be formed. Also, calculate the shock angle, pressure coefficient along the wedge and drag coefficient. Compare the FLUENT results with the corresponding analytical results.

Go to Step 1: Pre-Analysis & Start-Up

Go to all FLUENT Learning Modules